Lecture Notes on Fluid Dynamics

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Ref: D. J. Acheson: Elementary Fluid Mechanics, §8.5

7.5 Cyclonic current forced by a swirling wind

Of practical interest is the case of nonuniform wind stress on the surface. As an extremely simplified model we consider a vortical wind stress over a large sea¹. See Figure 7.5.1.

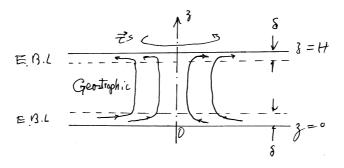


Figure 7.5.1: Steady cyclonic flow in a shallow sea forced by swirling wind

Let us restricting to a low Rossby number flow for simplicity. Continuity requires:

$$\frac{\partial u}{\partial x} + \frac{\partial v}{\partial y} + \frac{\partial w}{\partial z} = 0, \tag{7.5.1}$$

The momentum equations are

$$-fv = -\frac{1}{\rho} \frac{\partial p}{\partial x} + \nu \nabla^2 u \tag{7.5.2}$$

$$fu = -\frac{1}{\rho} \frac{\partial p}{\partial y} + \nu \nabla^2 v \tag{7.5.3}$$

$$0 = -\frac{1}{\rho} \frac{\partial p}{\partial z} + \nu \nabla^2 w \tag{7.5.4}$$

¹Acheson demonstrated a very similar problem of a circular layer of water bounded above and below by two horizontal planes. While the bottom plane rotates about the vertical axis at the rate Ω the top cover rotates steadily at a different rate $(1 + \epsilon)\Omega$.

The boundary conditions are: no slip on the bottom:

$$u = v = w = 0, \quad z = 0$$
 (7.5.5)

and given wind stress on the top:

$$\tau_{\theta z}^{S} = \rho Tr/2, \quad \tau_{rz}^{S} = 0, \quad z = H.$$
 (7.5.6)

The wind stress is cyclonic, where T is the curl of the wind sress vector:

$$\nabla \times \vec{\tau}^S = \vec{k} \left(\frac{1}{r} \frac{\partial}{\partial r} (r \tau_{\theta z}^S - \frac{1}{r} \frac{\partial \tau_{rz}^S}{\partial \theta}) \right) = \rho T \vec{k}. \tag{7.5.7}$$

In cartesian coordinates the wind stress components are:

$$\tau_{xz}^S = -\tau_{\theta z}^S \sin \theta = -\frac{\rho T}{2} r \sin \theta = -\frac{\rho T}{2} y, \tag{7.5.8}$$

$$\tau_{yz}^{S} = \tau_{\theta z}^{S} \cos \theta = \frac{\rho T}{2} r \cos \theta = \frac{\rho T}{2} x, \tag{7.5.9}$$

Kinematically we assume that

$$w = 0, \quad z = H.$$
 (7.5.10)

7.5.1 Inviscid core

Outside the surface an bottom boundary layers, we have

$$-fv_I = -\frac{1}{\rho} \frac{\partial p}{\partial x} \tag{7.5.11}$$

$$fu_I = -\frac{1}{\rho} \frac{\partial p}{\partial y} \tag{7.5.12}$$

This is clearly the state of geostrophyic balance. Momentum balence in the vertical direction is trivial,

$$0 = -\frac{1}{\rho} \frac{\partial p}{\partial z}$$

Consequently u_I and v_I must be independent of z. in accordance with the Taylor-Proudman theorem. Note that conservation of mass is automatically satisfied,

$$\frac{\partial u_I}{\partial x} + \frac{\partial v_I}{\partial y} = 0$$

and the vorticity is

$$\frac{\partial v_I}{\partial x} - \frac{\partial u_I}{\partial y} = -\frac{1}{f} \left(\frac{\partial^2 p}{\partial x^2} + \frac{\partial^2 p}{\partial y^2} \right)$$

The horizontal components $u_I(x, y), v_I(x, y)$ are not determined yet. The vertical velocity w_I can at best be a contant in z.

7.5.2 Bottom boundary layer

Let us keep the dominant viscous stress terms in the momentum equations,

$$-f(v-v_I) = \nu \frac{\partial^2 (u-u_I)}{\partial z^2}$$
 (7.5.13)

$$f(u - u_I) = \nu \frac{\partial^2 (v - v_I)}{\partial z^2}$$
 (7.5.14)

The boundary conditions are

$$u - u_I = -u_I \quad v - v_I = -v_I \quad z = 0$$

$$u - u_I \to 0 \quad v - v_I \to 0 \quad z \gg \delta$$

where

$$\delta = \sqrt{\frac{2\nu}{f}} \tag{7.5.15}$$

is the Ekman boundary layer thickness.

The solution is left to the reader as an exercise

$$u - u_I = -e^{-z/\delta} \left(u_I \cos \frac{z}{\delta} + v_I \sin \frac{z}{\delta} \right) \tag{7.5.16}$$

$$v - v_I = -e^{-z/\delta} \left(v_I \cos \frac{z}{\delta} - u_I \cos \frac{z}{\delta} \right). \tag{7.5.17}$$

From continuity, the vertical component can be computed. Let $\zeta = z/\delta$,

$$\frac{\partial w}{\partial z} = \frac{1}{\delta} \frac{\partial w}{\partial \zeta} = -\left(\frac{\partial u}{\partial x} + \frac{\partial v}{\partial y}\right)
= \left(\frac{\partial v_I}{\partial x} - \frac{\partial u_I}{\partial y}\right) e^{-\zeta} \sin \zeta + \left(\frac{\partial u_I}{\partial x} + \frac{\partial v_I}{\partial y}\right) \left(e^{-\zeta} \cos \zeta\right).$$
(7.5.18)

The second term vanishes, hence,

$$w = \delta \int_0^{\zeta} d\zeta \left(\frac{\partial v_I}{\partial x} - \frac{\partial u_I}{\partial y} \right) e^{-\zeta} \sin \zeta$$

$$= \delta \left(\frac{\partial v_I}{\partial x} - \frac{\partial u_I}{\partial y} \right) \frac{e^{-\zeta}}{2} \left(-\sin \zeta - \cos \zeta \right) \Big|_0^{\zeta}$$

$$= \frac{\delta}{2} \left(\frac{\partial v_I}{\partial x} - \frac{\partial u_I}{\partial y} \right) \left[1 - e^{-\zeta} \left(\cos \zeta + \sin \zeta \right) \right].$$

At the outer edge of the bottom boundary layer, $\zeta = z/\delta \gg 1$

$$w(\infty) \equiv \frac{\delta}{2} \left(\frac{\partial v_I}{\partial x} - \frac{\partial u_I}{\partial y} \right) = \frac{\delta}{2} \omega_I \tag{7.5.19}$$

where ω_I is the vorticity in the geostrophic interior. Thus there is vertical flux from the bottom boundary layer when the interior flow is horizontally nonuniform; this is called the Ekman pumping!

We still don't know the geostrophic flow field.

7.5.3 Surface boundary layer

The momentum equations are

$$-f(v - v_I) = \nu \frac{\partial^2 (u - u_I)}{\partial z^2}$$

$$f(u - u_I) = \nu \frac{\partial^2 (v - v_I)}{\partial z^2}.$$
(7.5.20)

On z = H the boundary conditions are

$$\nu \frac{\partial u}{\partial z} = -\frac{T}{2}y, \quad \nu \frac{\partial v}{\partial z} = \frac{T}{2}x, \quad z = H$$
 (7.5.21)

Far beneath the surface

$$u \to u_I, \quad v \to v_I; \quad (H - z) \gg \delta$$
 (7.5.22)

Let us introduce the boundary-layer coordinate

$$\eta = \frac{H - z}{\delta} \qquad 0 < \eta < \infty. \tag{7.5.23}$$

so that

$$\frac{\partial}{\partial z} \to -\frac{1}{\delta} \frac{\partial}{\partial \eta} \tag{7.5.24}$$

The solution satisfies the momentum equations and (7.5.22) is of the form

$$u - u_I = e^{-\eta} \left(A \cos \eta + B \sin \eta \right) \tag{7.5.25}$$

$$v - v_I = e^{-\eta} (B \cos \eta - A \sin \eta).$$
 (7.5.26)

In order to satisfy (7.5.21), we first note that

$$\frac{\partial u}{\partial \eta} = e^{-\eta} \left((-A + B) \cos \eta + (-A - B) \sin \eta \right) \tag{7.5.27}$$

$$\frac{\partial v}{\partial \eta} = e^{-\eta} \left((-A - B) \cos \eta + (A - B) \sin \eta \right). \tag{7.5.28}$$

Applying (7.5.21), we get

$$-\frac{\nu}{\delta}(-A+B) = -\frac{Ty}{2}, \quad -\frac{\nu}{\delta}(-A-B) = \frac{Tx}{2}$$
 (7.5.29)

with the results,

$$A = \frac{T\delta}{4\nu}(x - y), \quad B = \frac{T\delta}{4\nu}(x + y) \tag{7.5.30}$$

Hence the horizontal velocities are

$$u - u_I = \frac{T\delta}{4\nu} e^{-\eta} ((x - y)\cos \eta + (x + y)\sin \eta)$$
 (7.5.31)

$$v - v_I = \frac{T\delta}{4\nu} e^{-\eta} ((x+y)\cos \eta - (x-y)\sin \eta).$$
 (7.5.32)

By continuity

$$\frac{\partial w}{\partial z} = -\frac{1}{\delta} \frac{\partial w}{\partial \eta} = -\left(\frac{\partial u}{\partial x} + \frac{\partial u}{\partial y}\right)$$
$$= \frac{T\delta}{4\nu} e^{-\eta} (2\cos\eta + 2\sin\eta)$$

the vertrical velocity can be found,

$$w(\eta) = \frac{T\delta}{2\nu} \int_0^{\eta} d\eta \, e^{-\eta} \left(\cos \eta + \sin \eta\right)$$

$$= \frac{T\delta}{2\nu} \left[e^{-\eta} \left(-\cos \eta + \sin \eta \right) + e^{-\eta} \left(-\cos \eta - \sin \eta \right) \right]$$

$$= \frac{T\delta}{2\nu} \left[\left(1 - e^{-\eta} \cos \eta \right) \right]$$
(7.5.33)

At the outer edge of the surface boundary layer $\eta\gg 1$

$$w(\infty) = w_T = \frac{T\delta}{2\nu} \tag{7.5.34}$$

By Taylor-Proudman theorem, $w(z) = w_B = w_T$. Therefore

$$w_B = \frac{\delta}{2} \,\omega_I = \frac{T\delta}{2\nu} = w_T \tag{7.5.35}$$

and the interior vorticity is

$$\omega_I = \frac{T}{\nu}.\tag{7.5.36}$$

What are u_I and v_I ? In cylindrical polar coordinates

$$\omega_{I} = \frac{1}{r} \frac{\partial}{\partial r} (r u_{I_{\theta}}) - \frac{1}{r} \frac{\partial u_{I_{r}}}{\partial \theta} = \frac{1}{r} \frac{\partial}{\partial r} (r u_{I_{\theta}}).$$

Since $\partial/\partial\theta = 0$, we have,

$$\omega_I = \frac{1}{r} \frac{d}{dr} (r u_{I_{\theta}})$$
$$\frac{d}{dr} (r u_{I_{\theta}}) = \frac{T}{\nu} r$$

which implies

$$u_{I_{\theta}} = \frac{T}{2\nu} r.$$

Since

$$\frac{1}{r}\frac{\partial}{\partial r}\left(r\,u_{I_r}\right) + \frac{1}{r}\frac{\partial u_{\theta}}{\partial \theta} = 0$$

which leads to

$$u_{I_r} = 0.$$

The interior flow is geostrophic and cyclonic.

In cartesian form we have

$$u_I = -u_{I_{\theta}} \sin \theta = -\frac{T}{2\nu} r \sin \theta, \qquad (7.5.37)$$

$$v_I = u_{I_\theta} \cos \theta = \frac{T}{2\nu} r \cos \theta \tag{7.5.38}$$

Now the radial component inside the bottom boundary layer is

$$u_r = u_r - u_{I_r}$$

since $u_{I_r} = 0$. The latter is

$$u_r - u_{I_r} = -e^{-\zeta} \left[(u_I \cos \zeta + v_I \sin \zeta) \cos \theta + (v_I \cos \zeta - u_I \sin \zeta) \sin \theta \right]$$

$$= -e^{-\zeta} \left[\cos \zeta (u_I \cos \theta + v_I \sin \theta) + \sin \zeta (v_I \cos \theta - u_I \sin \theta) \right]$$

$$= -e^{-\zeta} \sin \zeta (v_I \cos \theta - u_I \sin \theta)$$

$$= -\frac{Tr}{2\nu} e^{-\zeta} \sin \zeta (\cos^2 \theta + \sin^2 \theta)$$

$$= -\frac{Tr}{2\nu} e^{-\zeta} \sin \zeta$$

and is negative in most of the boundary layer. Hence the flow spirals inward towards the z axis in the bottom boundary layer. Similarly one can show that the flow in the surface boundary layer has an outward radial component.

In summary, the swirling wind induces a vorticity T/ν in the geostrophic interior. The flow in the bottom Ekman layer spirals inward, rises vertically at a uniform velocity while spiralling at the angular velocity T/ν and maintaining a constant vorticity in the geotrophic interior, then spirals outward in the surface Ekman layer. The flow is therefore cyclonic.