Topics in Machine Elements



















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Critical speed of a shaft with a mass – synchronous case (slowest)



At $\omega' = 1$, no

At $\omega = 1$, no steady-state configuration; deflection grows linearly with time!

Shaft has stiffness k at the location of the flywheel, $\boldsymbol{\epsilon}$ is eccentricity

In steady-state,

$$kx = m\omega^{2}(x+\varepsilon) \rightarrow$$

$$(k - m\omega^{2})x = m\omega^{2}\varepsilon \rightarrow$$

$$x / \varepsilon = m\omega^{2} / (k - m\omega^{2})$$

$$= \omega^{2} / (1 - \omega^{2}) \quad \text{where } \omega' = \omega / \omega_{m}$$

ADDITIONAL formulas available for multiple masses on a shaft

Circular shafts in **combined** loading



Polar moment of inertia: Bending moment of inertia: Torque: Angular deflection

Bending shear stress: Bending moment: Max shear stress (at shaft surface): Max bending stress (at top/bottom surface): $\sigma_{h}(x) = M(x)r / I = 32 M(x) / \pi d^{3}$ Transverse shear stress (on u/d centerline): $\sigma_t(x) \sim 16 V(x) / 3 \pi d^2$

... and don't forget stress concentration!

 $J = \pi d^4 / 32$ $I = \pi d^4 / 64 \quad (= J / 2)$ Т $\phi = T L / J G$

V(x)dM(x)/dx = V(x) $\tau_{max} = Tr/J = 16T / \pi d^3$

Just what you need – Mohr stress!



$$s_{1} = (\sigma_{x} + \sigma_{y}) / 2$$

$$s_{2} = (\sigma_{x} - \sigma_{y}) / 2$$

$$\tau_{max} = sqrt(s_{2}^{2} + \tau_{xy}^{2})$$

$$tan 2\theta = \tau_{xy} / s_{2}$$

 $\tan 2\phi = \tau_{xy} / \sigma_2$ $\tan 2\phi = -\sigma_2 / \tau_{xy}$

 $\sigma_1 = s_1 + \tau_{max}$ $\sigma_2 = s_1 - \tau_{max}$

principal stresses

Fatigue of Engineering Materials

Basquin's equation: $\mathbf{L} = \mathbf{k}_{\mathbf{b}} \sigma^{-\mathbf{b}}$ where L is the lifetime (cycles) at stress level σ k_b, b are determined from test data Curve fits data for cycles at a given, constant stress level



Figure by MIT OpenCourseWare. Adapted from Fig. 6-11 in Shigley & Mischke.

For loading at multiple levels, Consider Miner's equation:

 $N_1/L_1 + N_2/L_2 + \dots = 1$

where

 $\begin{array}{l} \text{L}_{i} \text{ is the lifetime at stress level } \sigma_{i} \\ \text{N}_{i} \text{ is the number of cycles actually} \\ \text{ executed at } \sigma_{i} \end{array}$

Concept: Accumulation of damage

Simplified approach does not take into account the sequence of loading levels



q: twice the standard deviation of the underlying Gaussian stress

Four-Bar Linkage Kinematics



I = length of longest link

- s = length of shortest link
- p,q = lengths of intermediate links

Grashof's theorem:

If s+l <= p + q, then at least one link will <u>revolve</u>. If s+l > p + q, then all three links are <u>rocking</u>.



Categories l+s < p+ql + sl + sl + s = p + ql + s > p + q

double-crank, if s is the frame rocker-crank, if s is one side double rocker, if s is coupler change point triple-rocker

Let ϕ_i be the absolute angle of link r_i vector as shown The chain satisfies:

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X-loop: \mathbf{r}_1 \cos \phi_1 + \mathbf{r}_2 \cos \phi_2 + \mathbf{r}_3 \cos \phi_3 + \mathbf{r}_4 = \mathbf{0} (note \phi_4 = 0)
Y-loop: r_1 \sin \phi_1 + r_2 \sin \phi_2 + r_3 \sin \phi_3
                                                                                 = 0
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Two equations, two unknowns $[\phi_2, \phi_3]$ if ϕ_1 given – use a nonlinear solver



Courtesy of Alex Slocum. Used with permission.

B. Paul, Kinematics and dynamics of planar machinery, 1984.



Figure by MIT OpenCourseWare. Adapted from Fig. 1.51-1 in Paul, Burton. *Kinematics and Dynamics of Planar Machinery*. Englewood Cliffs, NJ: Prentice-Hall, 1979.

Slider-Crank Kinematics



X-loop: $\mathbf{r} \cos\theta - \mathbf{L} \cos\phi - \mathbf{s} = \mathbf{0}$ Y-loop: $\mathbf{r} \sin\theta - \mathbf{L} \sin\phi - \mathbf{e} = \mathbf{0}$ Two equations, two unknowns [s, ϕ] if θ is given

$$s_{max} = s_1 = sqrt [(L + r)^2 - e^2]$$

 $s_{min} = s_2 = sqrt [(L - r)^2 - e^2]$

 $\theta \text{ at } s_{max} = \theta_1 = \arcsin (e / (L + r))$ $\theta \text{ at } s_{min} = \theta_2 = \pi + \arcsin (e / (L - r))$

Slider moves to the right $s_{min} \rightarrow s_{max} : \theta_2 \rightarrow \theta_1$ Slider moves to the left $s_{max} \rightarrow s_{min} : \theta_1 \rightarrow \theta_2$



So time ratio TR = $(\theta_2 - \theta_1) / (2\pi - \theta_2 + \theta_1)$: captures "quick-return" characteristic



Figure by MIT OpenCourseWare. Adapted from Fig. 1.42-1 in Paul, Burton. *Kinematics and Dynamics of Planar Machinery*. Englewood Cliffs, NJ: Prentice-Hall, 1979.

Radial Ball Bearings

Ball Bearings in radial loading

- Load rating is based on fatigue:
 - Basic Rating Load C causes failure in 10% of bearings at 1 million cycles
- Hardness and finish of balls and rollers is critical!
 - Use e.g., high-carbon chromium steel 52100, min 58 Rockwell.
 - Finish balls to 50nm typical, races to 150nm typical
 - Quality indexed by ABEC rating: 1 to 9
- Examples of Ratings:

 #102: 15mm bore, 9x32mm dia: 	4.3 kN C	2.4 kN static
– #108: 40mm bore, 15x68mm dia:	13.6 kN C	10.9 kN static
– #314: 70mm bore, 20x110mm dia:	80 kN C	59 kN static

- Note static load rating < dynamic load rating!
- Scaling: life goes as load cubed
 - Decreasing the load by 1/2 will increase expected life by 8-fold, etc.

Effect of Axial Loading on Radial Bearings: Equivalent radial load

 $Max(1.2P_r, 1.2XP_r + YP_a)$

where P_r and P_a are axial and radial loads, and X,Y \rightarrow

Service factor $C_1 = [1 - 3+]$ to account for shock loads: $Max(1.2C_1P_r, 1.2C_1XP_r + C_1YP_a)$

Concept of <u>accumulated damage</u> (Miner's equation) applies

Use tapered roller bearings for large combined loads OR Radial bearings and thrust bearings separately



25 0.56 2.3 2.0 50 0.56 1.7 0.56 100 0.56 200 1.5 1.2 500 0.56 1000 0.56 1.0

Х

Y

 P_{a}/ZiD^{2}

Z = number of balls i = number of rings D = ball diameter

Confidence levels adjustment to lifetime:

90%	1.0
95%	0.62
99%	0.21

Helical Springs

Yes, you can derive the stiffness in a helical spring!

Ν

Ρ

 $L \sim \pi D N$

 $A = \pi d^2 / 4$

 $J = \pi d^4 / 32$

Let
Number of coils
Wire length
Wire area
Rotary MOI of wire
Axial load
Wire torsion from load

Wire torsion from loadT = P D / 2Torsional shear at wire surface τ_T Transverse shear at mid-line τ_t Total shear stress $\tau = \tau_T$

$$\begin{split} \tau_{T} &= T \; d \; / \; 2 \; J = 8 \; P \; D \; / \; \pi \; d^{3} \text{, and} \\ \tau_{t} &= 1.23 \; P \; / \; A = (0.615/c) \; x \; \tau_{t} \text{, so} \\ \tau &= \tau_{t} + \tau_{T} = (1 \; + \; 0.615/c) \; x \; \tau_{t} \\ & (\text{but } 0.615/c \; \text{is small if c is big}) \end{split}$$

Ρ

D/2

Straight-bar

d

equivalent to the helical spring

Ρ

Differential angle Differential deflection Integrated deflection Stiffness $\begin{array}{l} \delta \varphi = T \; \delta L \; / \; J \; G = 16 \; P \; c^2 \; \delta N \; / \; d^2 \; G \\ \delta x = \delta \varphi \; D \; / \; 2 \; \; (90 \; degrees \; away) \; \sim \; 8 \; P \; c^3 \; \delta N \; / \; d \; G \\ x = 8 \; P \; c^3 \; N \; / \; d \; G \\ \textbf{k} = \textbf{P} \; / \; \textbf{x} = \textbf{G} \; \textbf{d} \; / \; \textbf{8} \; \textbf{c}^3 \; \textbf{N} \end{array}$

c = D/d = coil diameter / wire diameter

Belleville Spring





Useful in assembly operations...

Spur Gears

Generation of Involute Teeth

Kinematic compatibility for friction cylinders: $r_1\omega_1 = r_2\omega_2$

Fundamental Law of Gears:

If the velocity of the driving gear is constant, so is the velocity of the driven gear *Fundamental Law dictates certain tooth shapes!*

Example of Involute gear teeth \rightarrow Cycloidal teeth also satisfy Fund. Law

Rolling contact when interface is between gear centers, otherwise sliding contact

Load is always applied along AB – so actual loading is the power transfer load, amplified by $1/\cos\phi$





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Epicyclic/Planetary Gearing!

Angle α on the power side (crank):

leads to rotation of the planet by $-\alpha N_2/N_1$ and rotation of the crank arm by α

The planet rotation <u>alone</u> (fix the crank angle to zero) drives the output shaft through an angle

$$(N_3/N_4)x(-\alpha N_2/N_1) = -\alpha N_3 N_2/N_4 N_1$$

while

the crank rotation <u>alone</u> (fix the planet angle to zero) rotates the output shaft by α

The net gear ratio is

 $\omega_{\text{load}}/\omega_{\text{power}} = 1 - N_3 N_2 / N_1 N_4$



Super-compact form

Because slight variations between N_2 and N_4 , and N_1 and N_3 , are easy to achieve, very high reductions are possible in a single stage, e.g., 100:1

Image sources

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