Using K_T and K_Q for design

these notes are landscape as plots are usually shown in that mode

we have seen in general the development of the Wageningen B series. The performance curves are available either in chart form or can be generated from polynomials:

▶ regression coeff. Re=2*10^6

polynomial representation

use in design

A typical design problem calls for designing a propeller that will provide the required thrust at a given speed of advance. These parameters result from applying thrust deduction and wake fraction to resistance and ship velocity respectively. Design will imply selecting a P/D from a B-series plot that will maximize open water efficiency.

For now we will arbitrarily pick a number of blades and expanded area ratio. Later we will address the criteria in their selection. Reviewing the non-dimensional forms of the parameters associated with thrust and speed:

$$K_{\rm T} = \frac{T}{\rho \cdot n^2 \cdot D^4} \qquad J = \frac{V_{\rm A}}{n \cdot D}$$

we have independent variables n and D. Normally one of these is determined by other criteria, e.g. maximum diameter by hull form, or n by the propulsion train design, so we will look at two cases, one in which D is fixed - determine n, and the other where n is fixed determine D

<u>case 1</u> given: V_A, T, D find n and P/D for maximum efficiency

only thing unknown is n, eliminate ... from ratio of K_{τ} and J

$$\frac{K_{t}}{J^{2}} = \frac{T}{\rho \cdot n^{2} \cdot D^{4}} \cdot \frac{n^{2} \cdot D^{2}}{V_{A}^{2}} = \frac{T}{\rho \cdot D^{2} \cdot V_{A}^{2}}$$

this says that propeller (full scale and model) must match this ratio which is a constant determined by T, V_A, D and ρ

 $Kt_over_J_sq := \frac{T}{\rho \cdot D^2 \cdot V_A^2}$

we can plot a curve of K_T vs J^2 and determine the points (values of J) for which K_T vs J for a given P/D match.

the design point for a particular propeller (B.n.nn) i.e. n is determined from the value of J that satisfies: $Kt(J) = constant J^2$

for example, let
$$Kt_over_J_sq := 0.544$$
what n i.e. J will satisfy the relationship for a B 5.75 propeller with P/D -1.0 $Kt_design(J) := Kt_over_J_sq \cdot J^2$ select using B_series $Z := 5$ $EAR := 0.75$ P_over_D := 1.0

determine intersection





busy plot of Kt, Kq, η_{0} and Kt = constant * J^2. see breakdown below. P/D not labeled but ~ J at Kt = 0

intersection solution

plot with only Kt but vertical lines at J for Kt/J^2 = Kt to show points which satisfy the design requirements



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note the η_0 at each J intersection and select the maximum (P/D curves not well labeled, P/D ~ = J at K_T=0. left to right lowest to highest

only thing unknown is D, eliminate ... from ratio of K_{T} and J

$$\frac{K_{t}}{J^{4}} = \frac{T}{\rho \cdot n^{2} \cdot D^{4}} \cdot \frac{n^{4} \cdot D^{4}}{V_{A}^{4}} = \frac{T}{\rho} \cdot \frac{n^{2}}{V_{A}^{4}}$$

 $Kt_over_J_4 := \frac{T}{\rho \cdot D^2 \cdot V_A^2}$

this says that propeller (full scale and model) must match this ratio which is a constant determined by T, $V_{\text{A}},$ n and ρ

we can plot a curve of K_T vs J⁴ and determine the points (values of J) for which K_T vs J for a given P/D match.

for example, let

 $Kt_over_J_4 := 0.544$

$$\underbrace{\text{Kt} \text{design}(J) := \text{Kt}_{\text{over}} \text{J}_{4} \cdot \text{J}^{4}}_{\text{Select using B}_{\text{series}}} = 5 \qquad \underbrace{\text{EAR}}_{\text{Z}:= 5} = 0.75$$

the design point for a particular propeller (B.n.nn) i.e. n is determined from the value of J that satisfies: $Kt(J) = constant J^4$

since the process is identical to case 1, only the final result is shown

intersection solution



this case appears to have maximum at

 $J_{ans} = 0.74$ $P_{over}_{D_{ans}} = 1$

 $\eta(J_{ans}, EAR, z, P_{over}D_{ans}) = 0.67$

and ...
$$D = \frac{V_A}{J_{ans \cdot n}}$$
 where V_A and n are known as described above

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